- Overview of parachute sizes and mass for given entry mass
- Design drivers
- Design cases
- Materials
- Planform
- Testing

Design Drivers

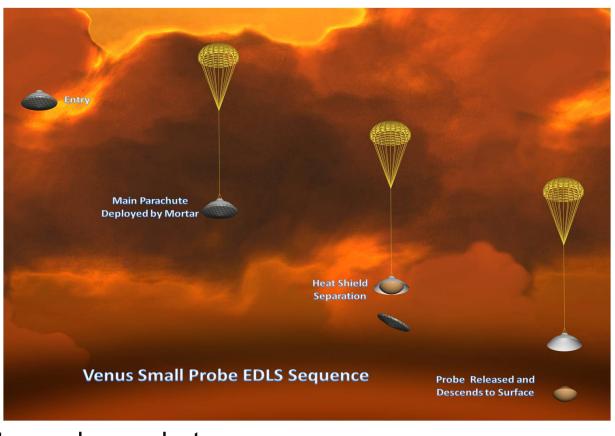


- Highly Corrosive Atmosphere
 - Sulfuric Acid
 - Drives material selection
- Similar Atmospheric Density at Deployment Altitude to Earth
 - At 50 km, 1 Bar, 75°C
 - Simplifies testing
- Rate of Descent
 - Typically at a proscribed altitude, not at landing due to thick atmosphere
- Deployment
 - Mortar deployed parachute to push parachute through base region of reverse flow behind vehicle

Small Probe Sequence

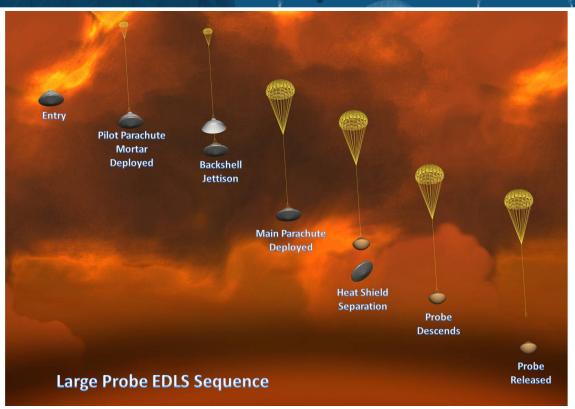
- 100 kg entry mass
- Mortar deployment of main parachute
- Heat shield separation after stabilization
- Release probe at required point
- 1.8 m parachute





Large Probe Sequence

- 500 kg entry mass
- Mortar deployment of pilot parachute
- Backshell released immediately following pilot mortar fire
- Pilot separates backshell, extracts main parachute

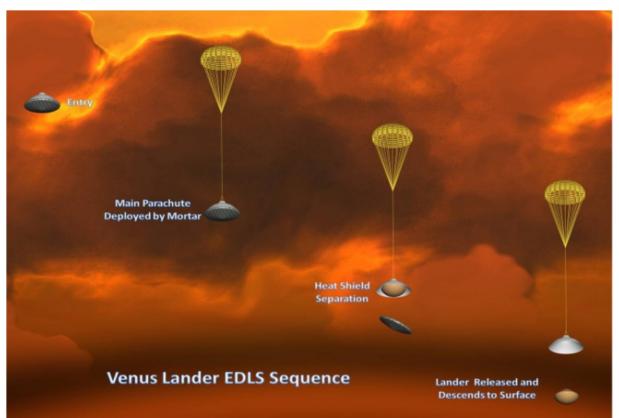


- Heat shield separation after stabilization
- Release probe at required point
- 5.0 m parachute, 7 kg total system mass



Lander Sequence

- 1,000 kg entry mass
- Mortar deployment of main parachute
- Heat shield separation after stabilization
- Release lander at required point
- 5.0 m parachute

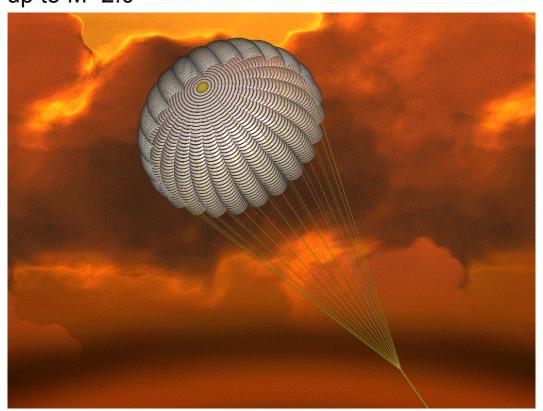


- 11 kg total mass, mortar and parachute
 - Higher mass due primarily to larger mortar



Other Options

- Higher deployment Q
 - Typical Venus mission deploy M=0.80
 - Conical ribbons deploy reliably up to M=2.0
- Larger parachute
 - Conical ribbons over 12 m have been flown
 - Allows ~2,000 kg entry mass





- Sulfuric acid clouds at parachute deployment altitudes
 - 50 to 62 km
 - 80% concentration at 80°C
- Pioneer Venus used Dacron[®]
- Current choice fabric is Vectran[®]
 - High tenacity liquid crystal polymer
 - Very resistant to sulfuric acid
 - Has space heritage
 - Airbags for Pathfinder, Mars Exploration Rovers
 - Meets total mass loss and collected volatile condensable mass requirements

- Pioneer Venus used ribless guide surface drogue and conical ribbon main
 - Intended architecture was RGS for both
 - 1.37 m RGS pilot performed well in testing
 - 6.74 m RGS main demonstrated inflation and equilibrium problems
 - Substituted 4.5 m conical ribbon
- Use heritage RGS pilot for small probe, or large probe pilot
- Use heritage conical ribbon main for lander or large probe main
- Investigate other options such as variable porosity conical ribbon if need to qualify a new parachute

Full Scale Testing



- Qualify components individually, then complete system end-to-end flight test
- Recent and current Mars missions all use Viking data from tests in early 1970's
- Leverage data from Pioneer Venus and extensive parachute testing conducted in 1960's and 1970's
- Two options for end to end testing
 - High altitude balloon drop
 - Sacrifices angle of attack and flight path angle for less expensive testing
 - Intermediate altitude balloon drop with rocket boost
 - · Achieves specific angle of attach and FPA

Scaled Testing

Airborne Systems

Until recently, scaled testing of ribbon parachutes not practical

NASA has successfully tested a 10% scale model of Orion capsule

drogue parachute (VPCR)

 Results compare favourably with drop test data from Orion test campaign

 Note that scaled testing can augment full size testing, not replace it.



10% Scale Conical Ribbon in Texas A&M Low Speed Tunnel (Photo courtesy of NASA)

